

Dr. Maverick's Boom Junk KC-135 WAG's

Power-Off - Gliding Distance (for LPU fit/preposition):

Note: R-Model WAG.

$$\text{Gliding Distance} = (\text{Altitude} * 15) / 6076$$

Example:

$$\text{Altitude} = 30,000 \text{ ft.}$$

$$30,000 * 15 / 6076 = 74.1 \text{ NM}$$

Take-Off Rotation Speed:

Note: Computes Vrot within a few knots.

$$\text{Vrot} = \text{Gross Weight} / 3 + 75 \text{ (30 Flaps)} \text{ or } + 77 \text{ (20 Flaps)}$$

Example:

$$\text{Gross Weight} = 225.0$$

$$225.0 / 3 + 77 = 152 \text{ KIAS}$$

Landing Approach Descent Rate:

Note: Computes approach descent rate for 3.0 degree glideslope.

$$\text{Approach Speed} * 10 / 2 = \text{Descent Rate (FPM)}$$

Example:

$$150 * 10 / 2 = 750 \text{ feet per minute}$$

Landing Approach Speed (50 Flaps):

Note: Computes 50 flap approach speed within a few knots.

$$\text{Approach Speed} = (((\text{Gross Weight} - 100) / 10) * 4) + 114$$

Example:

$$\text{Gross Weight} = 195.0$$

$$(((190.0 - 100) / 10) * 4) + 114 = 150 \text{ KIAS}$$

Easy Method:

Take the middle and last digit of the gross weight (i.e. 190 yields 9.0) multiply by 4, then add to 114.

CG WAG:

Note: Computes CG within a few tenths of a percent.

$$\text{CG} = 33 - \text{FWD} - 1/3\text{CTR} + 2/3(\text{AFT} + \text{RES})$$

Example:

$$\text{FWD: } 11.2$$

$$\text{CTR: } 0.0$$

$$\text{AFT: } 1.5$$

$$\text{RES: } 5.8$$

$$33 - 11.2 - 1/3(0) + 2/3(1.5 + 5.8) = 26.7 \% \text{ CG}$$

Ballast Fuel:

Notes: Computes approximate forward body fuel required to obtain a desired CG, reserves full, mains balance, center wing empty.

$$(((\text{OPMOM} + \text{RES} + \text{AFT}) / (\text{OPWT} + \text{RES} + \text{AFT}) * 1000) - \text{DCGA}) / 2.79$$

Example:

$$\text{Desired CG: } 25.0$$

$$\text{Desired CGA (DCGA): } 846.7((25 * 241.9 / 100) + 786.2)$$

OPMOM: 107,011

OPWT: 122,719

CTR: 0.0

AFT: 1.5M / 1,580

RES: 5.8M / 6,272

$((107011+1580+6272)/(122719+1500+5800))*1000 - 846.7 =$

$((114863 / 130019) * 1000) - 846.7) / 2.79 =$

13.2M forward (reserves full)